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Fifth Semester B.E. Degree Examination, May 2017 Aircraft Structure – I

Time: 3 hrs. Max. Marks: 100

Note: Answer any FIVE full questions, selecting atleast TWO questions from each part.

PART - A

- 1 a. List and discuss various loads acting on the structural components of aircraft. (04 Marks)
 - b. Define interia loads and derive the formula for minimum resultant inertial forces and total torque. (08 Marks)
 - c. Discuss the structural functions of various components with reference to wing with neat diagram. (04 Marks)
- 2 a. Sketch typical stress-strain curves for below mentioned material showing their important mechanical properties:
 - i) Aluminium ii) copper iii) brittle material iv) mild steel v) rubber. (05 Marks)
 - b. Briefly discuss aluminium alloys and steel as aircraft material with its area of applications.
 (09 Marks)
 - 2. Discuss importance of non-metals as aircraft materials with applications. (06 Marks)
- 3 a. Discuss the tensile properties of mild steel. Clearly mention the linear and non-linear mechanical properties. (10 Marks)
 - b. What is fatigue? List and explain fatigue properties of material. (06 Marks)
 - c. Define creep and discuss creep properties. (04 Marks)
- 4 a. Distinguish statically determinate and indeterminate structure with example. (06 Marks)
 - b. A truss load as shown in Fig.Q4(b). Determine the forces in all the members of the truss.

 (14 Marks)

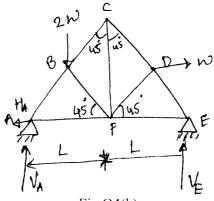
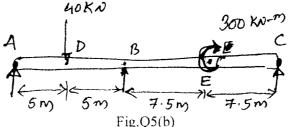


Fig.Q4(b)

PART - B

5 a. A continuous beam ABC of constant moment of inertia is simply supported at A, B, and C. The beam carries a central point load of 40 kN in span AB and a central clock wise couple of 300 kN-m in the span BC, as shown in Fig.Q5(a). Find the support moments and plot the shear force and bending moment diagram. (10 Marks)



b. State and prove Maxwell reciprocal theorem.

(10 Marks)

6 a. Show that critical load for long column is given by:

$$P = \frac{n^2 \pi^2 EI}{L^2}$$
, where $(n = 1, 2, 3 - - -)$. (10 Marks)

- b. A cast iron hallow column, having 100 mm external diameter and 80 mm internal diameter is used as a column of 2.4m length. Using Rankines formulas, Determine the crippling load, when both the ends are fixed. Take $f_c = 600 \text{ N/mm}^2$ and $a = \frac{1}{1600}$. (10 Marks)
- 7 a. Derive the equilibrium equation of 3-D elastic body, subjected to externally applied force system. Deduce the same to 2-D plane stress condition. (14 Marks)
 - b. Explain plane stress and plane strain conditions with example. (06 Marks)
- 8 a. Explain the need for failure theories and briefly describe maximum principal stress theory and distortion energy theory with their limitations and applicability. (12 Marks)
 - b. If the principal stress at point in an elastic material are 2f tensile, 1.5f tensile and f compressive. Calculate the value of f at failure according to any three failure theories. The elastic limit in simple tension is 210 N/mm^2 and $\mu = 0.3$. (08 Marks)

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